

CHAPTER - II

Approximate Solution of Steady Laminar
Flow Past a Horizontal Plate embedded
in a Saturated Porous Medium

1) Introduction :

The approximate solution of the velocity boundary layer based on Karman momentum integral equation was first obtained by E. Pohlhausen [3].

The exact solution of the velocity distribution in boundary layer flow past a flat plate was investigated by Blasius (1908) [5] and the corresponding numerical solution for the thermal boundary layer with and without frictional heat was studied by E. Pohlhausen. []

Recently B.C. Chandrasekhara [1] obtained solution for axial and transverse boundary layer equation in the case of steady laminar flow past a horizontal plate embedded in saturated porous medium.

In this problem we assume the governing equation of motion for two dimensional boundary layer following B.C. Chandrasekhara [1], Yekta and B.B. Waghmode [2], where the properties of the fluid and the porous medium such as viscosity permeability are constant and 6th degree velocity profile for the boundary layer flow over a horizontal plate embedded in a saturated porous medium. The velocity distribution is some function η of the ratio y/δ , δ being the boundary layer thickness. } 9

The aim of this analysis is to find the velocity distribution graphically and obtain the displacement thickness δ_1 momentum thickness δ_2 and shear stress at the wall τ_0 .

2) Mathematical Formulation :

For mathematical analysis we assume the governing equation of motion for two dimensional boundary layer flow following B.C.Chandrasekhara [1] and Yekta and B.B.Waghmode [2] where the properties of the fluid and the porous medium such as viscosity, permeability are constant.

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2} - \frac{\nu}{K} u \quad \dots (1)$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad \dots (2)$$

where K : permeability of the porous medium

ν : Kinematic viscosity of the fluid

The boundary conditions for this problem are

$$\begin{array}{l} y = 0 \quad v = u = 0 \\ y = \delta_2 \quad u = U_\infty, v = 0 \\ \frac{du}{dy} = 0 \quad \text{when } y = \delta_2 \end{array} \quad \dots (3)$$

3) Analysis of the Problem :

To solve the above equation we define a stream function

$$u = \frac{\partial \psi}{\partial y} \quad \text{and} \quad v = - \frac{\partial \psi}{\partial x} \quad \dots (4)$$

Substituting these values in (1) we get

$$\frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} = \nu \frac{\partial^3 \psi}{\partial y^3} - \frac{\nu}{\kappa} \frac{\partial \psi}{\partial y}$$

i.e. $\frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} = \nu \frac{\partial^3 \psi}{\partial y^3} - B \frac{\partial \psi}{\partial y} \quad \dots (5)$

$$\text{where } B = \nu / \kappa.$$

Introducing a similarity transformations as

$$\eta = \left(\sqrt{\frac{U_\infty}{\nu x}} \right) y \quad \therefore y = \eta \sqrt{\frac{\nu x}{U_\infty}} \quad \dots (6)$$

$$\psi(\eta) = \sqrt{\nu x U_\infty} f(\eta)$$

$$u = \frac{\partial \psi}{\partial y} = U_\infty f'(\eta)$$

$$u \frac{\partial u}{\partial x} = U_\infty^2 f'(\eta) \frac{\partial f'}{\partial x}$$

$$v = \sqrt{\nu U_\infty x} \frac{\partial f}{\partial x} + \sqrt{\frac{\nu U_\infty}{x}} \frac{f(\eta)}{2}$$

$$v \frac{\partial u}{\partial y} = U_{\infty}^2 f''(\eta) \frac{\partial f}{\partial x} + U_{\infty}^2 \frac{f(\eta) f''(\eta) x^{-1}}{2}$$

$$\frac{\partial^2 \psi}{\partial y^2} = U_{\infty} f''(\eta) \sqrt{\frac{U_{\infty}}{\gamma} x}$$

$$\therefore \frac{\partial^3 \psi}{\partial y^3} = \frac{U_{\infty}^2 f'''(\eta)}{\gamma x}$$

$$\therefore \gamma \frac{\partial^3 \psi}{\partial y^3} = \frac{U_{\infty}^2 f'''(\eta)}{x}$$

Putting all the above values in (1) we get

$$U_{\infty}^2 f' \frac{\partial f'}{\partial x} - U_{\infty}^2 f''(\eta) \frac{\partial f}{\partial x} - \frac{U_{\infty}^2 f(\eta) f''(\eta) x^{-1}}{2} = U_{\infty}^2 f'''(\eta) x^{-1} - BU_{\infty} f'(\eta) \quad \dots (7)$$

where $B = \gamma / K$.

repeating

Dividing both side by $U_{\infty}^2 x^{-1}$ equation (7) becomes

$$f'''(\eta) + \frac{1}{2} f(\eta) f''(\eta) - \left(\frac{Bx}{U_{\infty}} \right) f'(\eta) = 0$$

$$\text{i.e. } 2f'''(\eta) + f(\eta) f''(\eta) - 2Bf'(\eta) = 0 \quad \dots (8)$$

$$\text{where } S = \frac{Bx}{U_{\infty}} = \frac{\gamma x}{KU_{\infty}} = \frac{\gamma x^2}{KU_{\infty}x} = \frac{x^2}{\frac{KU_{\infty}x}{\gamma}} = \frac{x^2}{KRe}$$

where $\text{Re } \frac{U_{\infty} x}{\delta}$

with the Boundary conditions

$$\left. \begin{aligned} \eta = 0, \quad u = 0, \quad f = 0, \quad f' = 0 \\ \eta = \infty, \quad u = U_{\infty}, \quad f' = 1 \end{aligned} \right\} \dots (9)$$

The solution of non-linear equation (8)

$$\frac{u}{U_{\infty}} = f(\eta) = \sum_{i=0}^6 a_i \eta^i \dots (10)$$

$$\frac{u}{U_{\infty}} = 1 \quad \text{for } \eta \gg 1 \quad \text{where } \eta = y/\delta$$

We have to determine the coefficient $a_0, a_1, a_2, a_3, a_4, a_5$ and a_6 . We prescribe the following boundary and compatibility conditions

$$\left. \begin{aligned} y = 0; \quad u = 0, \quad \frac{\partial^2 u}{\partial y^2} = 0, \quad \frac{\partial^3 u}{\partial y^3} = 0 \\ y = \delta; \quad u = U_{\infty}, \quad \frac{\partial u}{\partial y} = \frac{\partial^2 u}{\partial y^2} = \frac{\partial^3 u}{\partial y^3} = 0 \end{aligned} \right\} \dots (11)$$

By using boundary conditions (11) we get

$$\left. \begin{aligned} a_0 = 0 \quad a_2 = 0 \quad a_3 = 0 \quad \text{and} \\ a_1 + a_4 + a_5 + a_6 = 1 \\ a_1 + 4a_4 + 5a_5 + 6a_6 = 0 \\ 12a_4 + 20a_5 + 30a_6 = 0 \\ 24a_4 + 60a_5 + 120a_6 = 0 \end{aligned} \right\} \dots (12)$$

(37)

Solving the above equations we get

$$a_1 = 2, a_4 = -5, a_6 = -2, a_5 = 6 \quad \dots (13)$$

Substituting the values of a_0 to a_6 in equation (10) we obtain the velocity in the form

$$\frac{u}{U_\infty} = f(\eta) = 2\eta - 5\eta^4 + 6\eta^5 - 2\eta^6 \quad \dots (14)$$

We calculate the displacement thickness δ_1 , momentum thickness δ_2 and shear stress at the wall (τ_0)

1) Displacement thickness (δ_1)

$$\delta_1 = \int_0^1 \left(1 - \frac{u}{U_\infty}\right) d\eta$$
$$\delta_1 = 0.2857 \delta \quad \dots (15)$$

2) Momentum thickness (δ_2)

$$\delta_2 = \int_0^1 \frac{u}{U_\infty} \left(1 - \frac{u}{U_\infty}\right) d\eta$$
$$\delta_2 = 0.7693 \delta \quad \dots (16)$$

3) Shearing stress at the wall (τ_0)

$$\tau_0 = \mu \frac{U_\infty}{\delta} \left(\frac{\partial f}{\partial \eta} \right)_{\eta=0}$$

(38)

$$\left(\frac{\partial f}{\partial \eta} \right)_{\eta=0} = 2$$

$$\tau'_0 = \frac{2\mu U_\infty}{\delta} \quad \dots (17)$$

The momentum integral equation may be solved easily if δ_2 instead of δ is regarded as known function and for this we write the Karman momentum equation as

$$\frac{U}{\delta_2} \frac{d\delta_2}{dx} + \left(2 + \frac{\delta_1}{\delta_2} \right) \frac{\delta_2^2}{\delta_2} \frac{dU}{dx} = \frac{\tau'_0 \delta_2}{\mu U} \quad \dots (18)$$

Substituting the values of δ_1 and δ_2 from (15) and (16) in equation (18) and simplifying we get

$$\delta = 0.0495 \sqrt{\frac{\nu x}{U_\infty}} \quad \dots (19)$$

From equation (19) equation (15), (16) and (17) will take following forms

$$\delta_1 = 1.7208 \sqrt{\frac{\nu x}{U_\infty}}$$

$$\delta_2 = 0.6672 \sqrt{\frac{\nu x}{U_\infty}}$$

$$\tau'_0 = 0.3306 \sqrt{\frac{\nu x}{U_\infty}}$$

(39)

Table

η	0	0.1	0.2	0.3	0.4
$f(\eta)$	0	0.199558	0.39379	0.572622	0.725248
η	0.5	0.6	0.7	0.8	0.9
$f(\eta)$	0.84375	0.925248	0.972622	0.993792	0.999558

From the table velocity distribution curve has been obtained.

4) Discussion and Conclusions :

- 1) As η increases from 0 to 1 the velocity increases.
- 2) Displacement thickness, Momentum thickness and shearing stress at the wall are calculated.

(40)

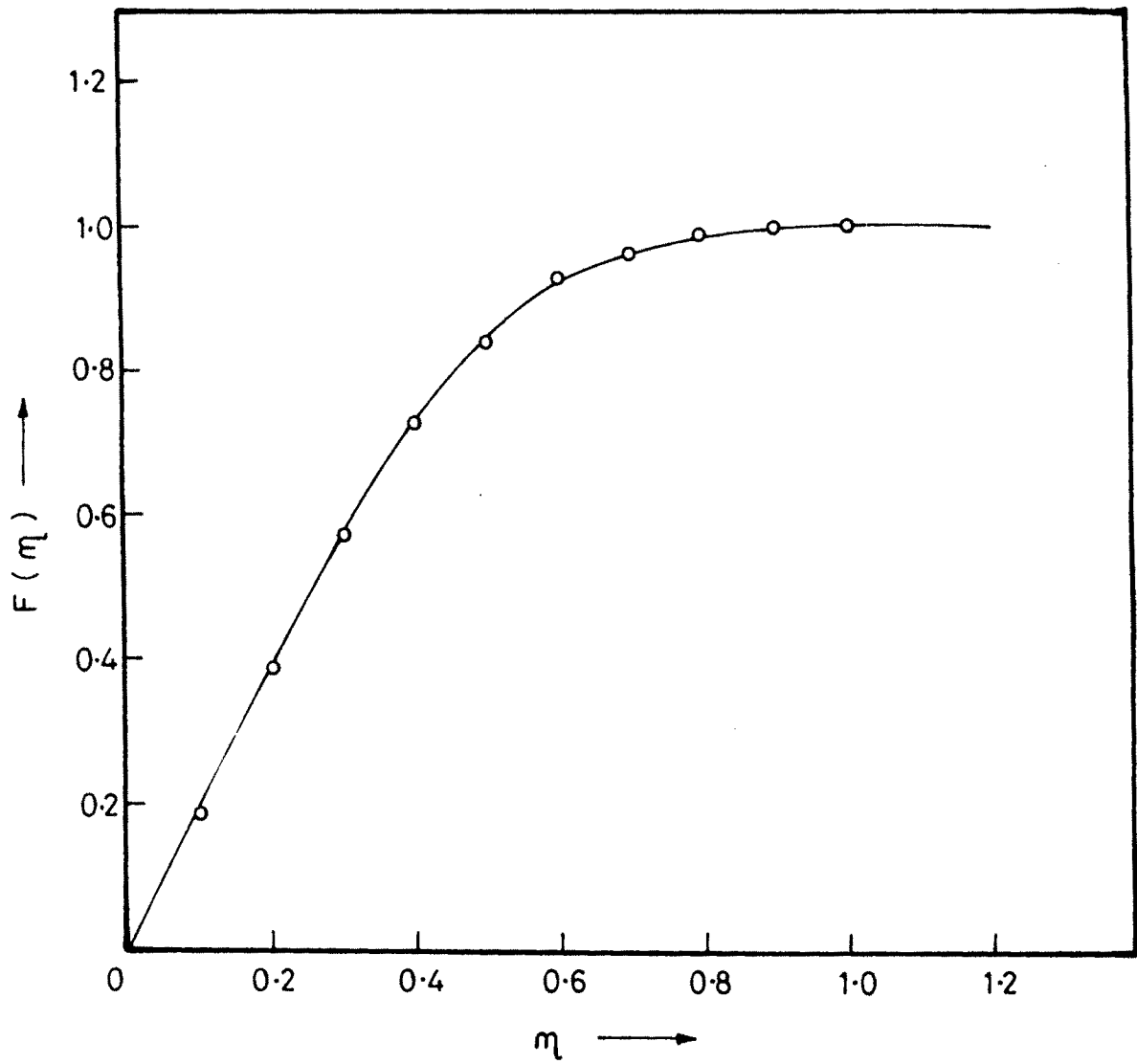


Fig.1 - VELOCITY DISTRIBUTION .

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